

ABSTRACT

A high-pressure turbine of a gas-turbine engine provided with a turbine disk (1) carrying rotor blades (2) and with stator blades (3), and with a lateral wheel cavity (5) being formed between the turbine disk (1) and the stator ring (4), wherein a seal (8) is provided in axial direction between stator blade platforms (6) and rotor blade platforms (7) and which is arranged radially outwardly adjacent a main gas duct (9).